## **RL1-90**



### **RLG SINGLE-AXIS INDEXING INERTIAL NAVIGATION SYSTEM**

#### **PRODUCT DESCRIPTION**

The RL1-90 Inertial Navigation System is a versatile and reliable navigation solution. It combines the Type 90 Ring Laser Gyroscope and Quartz Flexible Accelerometer to deliver accurate navigation information, including speed, position, and attitude. The system seamlessly integrates with auxiliary equipment such as GNSS, altimeters, and airspeed meters, catering to the needs of air and ground carriers.

This system finds wide applications across various carrier platforms, including tanks, armored vehicles, aircraft, drones, ships, automobiles, high-speed rail, unmanned vehicles, mobile satellite communication systems, and drilling operations. It offers precise control over flight, attitude stabilization, weapon stabilization platforms, positioning, and orientation.

With its advanced technology and robust design, the RL1-90 Inertial Navigation System is a reliable choice for meeting

#### **PRODUCT FEATURES**

- Sing-axis indexing mechanism for error mitigation
- High accuracy ring laser gyro and quartz accelerometer
- Optional static or moving base self-alignment
- Error parameters calibration and compensation in full temperature range

#### 🕈 APPLICATION AREAS

• Under-sea vehicle navigation

navigation requirements in diverse industries and applications.



- Optional diverse input interfaces for GNSS/Odometer/DVL
- Configurable navigation modes
- Excellent environmental suitability
- Military standards
- Positioning and north-finding for land vehicle
- Stabilization and control for moving carrier
- Attitude measurement for demanding applications

#### MAIN FUNCTIONS

- It has the function of outputting information such as carrier position, heading, attitude angle, angular rate and speed in real time;
- It has working modes such as pure inertial navigation and INS/GNSS (including Beidou) integrated navigation;
- Possess the function of receiving satellite navigation information provided by external time system frequency standard equipment;
- It has the function of ground self-alignment and supports the function of air alignment;
- It has functions such as power-on self-test, periodic self-test, status report, installation error compensation, and non-volatile storage.

#### **PERFORMANCE INDICATORS**

System accuracy System Accuracy	Pure Inertial Navigation/Pure Inertial Navigation	2.0nmile/3d, PEAK
	Integrated Navigation/Navigation with GNSS	≤5m , 1σ



# RL1-90

	Heading angle /Heading Horizontal attitude (roll and pitch ) Horizontal Attitude (roll & pitch)		0.01°, RMS
			0.005°, RMS
		ure Inertial Velocity	1.0 m/s , RMS
	GNSS Integrated navigation Velocity		0. 1 m/s , RMS
Indicators of inertial devices Gyro and Accelerometer Parameters	laser gyroscope _ Gyroscope	Range/Range	$\pm$ 6 00 deg/s
		Bias Stability	≤0.002 deg/h, 1σ
		Bias Repeatability	≤0.002 deg/h, 1σ
		Scale Factor non-linearity	1 ppm
	Accelerometer Accelerometer	Range/Range	± 15g
		Bias Stability	≤10μg , 1σ
		Zero bias repeatability Bias Repeatability	≤10µg,1σ
		Scale Factor non-linearity	15 ppm
<b>1</b>	Cold Start		$\leq$ 15 min
align time Alignment Time	Re-Start		$\leq 10 \min$
	Air/In-Flight Start		≤15min
Working hours Operation Time	Continuous working time/Operation Time		more than 10h
	Supply voltage/Voltage		18~36VDC
The state	Power Consumption		≤40W @ 24VDC
Interface Features interface	Elect	rical interface/Electrical	RS232 × 2 RS422 × 3 CAN × 2 Ethernet × 1 1pps × 1
	Data Update Rate (configurable)		200Hz@115.2kbps
	Operating Temperature		-40°C~+65°C
	Storage temperature/Storage Temperature		-55°C~+85°C
Use environment	Use Altitude/Altitude		20000m
Environmental	Humidity		≤95% (+25°C)
	Vibration/Vibration		5g @ 20~2000Hz
	Shock/Shock		40 g, 11 ms, 1/2 Sine
Physical properties	Dimensions/ Size ( $\Phi^*H$ )		$\leq$ 346 x 435 mm
Physical	Weight/ Weight		45 kg

Note: The structure can be customized according to the user's requirements.

### GYROSCPE MOUNTING DIMENSIONS

- The whole system is composed of two parts: the inertial navigation main instrument and the inertial navigation main instrument bracket.
- Among them, the external dimensions of the main instrument are as follows:



